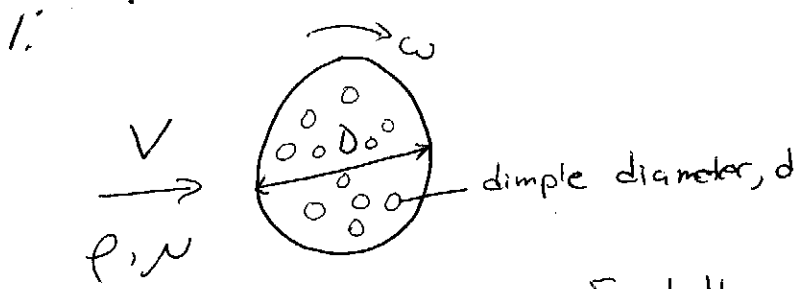


Key



Find the dimensionless torque, T
 { other π groups

$$T = (\rho, \nu, V, d, D, \omega)$$

$n = 7$ dimensional parameters

note angles are dimensionless

T	ρ	ν	V	d	D	ω
$\frac{ML^2}{T^2}$	$\frac{M}{L^3}$	$\frac{M}{LT}$	$\frac{L}{T}$	L	L	$\frac{1}{T}$

- Choose primary dimension system - $MLT \Rightarrow m = 3$

$\therefore r = 3$

- Dimensional matrix

	T	ρ	ν	V	d	D	ω
M	1	1	1	0	0	0	0
L	2	-3	-1	1	1	1	0
T	-2	0	-1	-1	0	0	-1

$$\begin{vmatrix} 1 & 1 & 0 \\ -3 & -1 & 1 \\ 0 & -1 & -1 \end{vmatrix} = (1)(1)(-1) - (1)(3-0) + 0() \neq 0$$

$\therefore \text{rank} = r = 3$

$\therefore n - r = 4$ π groups

* Select repeating variables: ρ, V, D

$$\pi_1 = \rho^a V^b D^c T = M^0 L^0 T^0$$

$$\left(\frac{M}{L^3}\right)^a \left(\frac{L}{T}\right)^b L^c \frac{ML^2}{T^2} = M^0 L^0 T^0$$

$M: a + 1 = 0 \Rightarrow a = -1$

$L: -3a + b + c + 2 = 0 \Rightarrow c = -3$

$T: -b - 2 = 0 \Rightarrow b = -2$

$$\Rightarrow \pi_1 = \frac{T}{\rho V^2 D^3}$$

1. Continued

(2)

$$\Pi_2 = \nu \rho^a V^b D^c = M^0 L^0 T^0$$
$$\frac{M}{L T} \left(\frac{M}{L^3}\right)^a \left(\frac{L}{T}\right)^b L^c$$

$$M: 1 + a = 0 \quad a = -1$$

$$L: -1 - 3a + b + c = 0 \Rightarrow c = 1 + 3(-1) + 1 = -1$$

$$T: -1 - b = 0 \Rightarrow b = -1$$

$$\therefore \Pi_2 = \frac{\nu}{\rho V D} = \frac{1}{Re_D}$$

from inspection:

$$\Pi_3 = \frac{d}{D}$$

$$\Pi_4 = \omega \rho^a V^b D^c = M^0 L^0 T^0$$
$$\left(\frac{1}{T}\right) \left(\frac{M}{L^3}\right)^a \left(\frac{L}{T}\right)^b L^c$$

$$M: a = 0$$

$$L: -3a + b + c = 0 \quad c = 1$$

$$T: -1 - b = 0 \Rightarrow b = -1$$

$$\Rightarrow \boxed{\Pi_4 = \frac{\omega D}{V}}$$

$$\textcircled{2} \quad \rho \left(u \frac{du}{dx} + v \frac{dv}{dy} \right) = - \frac{dp}{dx} + \mu \left(\frac{d^2u}{dx^2} + \frac{d^2v}{dy^2} \right)$$

A
B
C

a) If $A \sim 0$ what is non-dimensional pressure :

$$\frac{\partial p}{\partial x} = \mu \frac{d^2u}{dy^2}$$

$$u_x = \frac{u}{U} \quad x_x = \frac{x}{L}$$

$$p^* = \frac{p}{P_s}$$

$$\frac{1}{L} \frac{\partial p}{\partial x_x} = \mu \frac{U}{L^2} \frac{d^2u_x}{dy_x^2}$$

$$\frac{\partial p}{\partial x_x} \frac{L}{\mu U} = \frac{d^2u_x}{dy_x^2}$$

Hence, $p^* = \frac{p}{P_s} \Rightarrow P_s = \frac{\mu U}{L}$

(b) If $e \sim 0$

$$\rho \left(u \frac{du}{dx} \right) = - \frac{dp}{dx}$$

$$\rho \frac{U^2}{L} \left(u_x \frac{du_x}{dx_x} \right) = - \frac{1}{L} \frac{dp}{dx_x}$$

$$u_x \frac{du_x}{dx_x} = - \frac{dp}{dx_x} \frac{L}{\rho U^2} \Rightarrow P_s = \rho U^2$$

CHAPTER 8

3.

Exercise 8.1

Given

$$P = f(d, U, \omega, c, \rho, \mu)$$

Dimension of power is $[P] = [W]/T = FL/T = (MLT^{-2})L/T = ML^2/T^3$.

The dimensional matrix is

	P	d	U	ω	c	ρ	μ
M	1	0	0	0	0	1	1
L	2	1	1	0	1	-3	-1
T	-3	0	-1	-1	-1	0	-1

We find

$$\begin{vmatrix} 1 & 0 & 0 \\ 2 & 1 & 1 \\ -3 & 0 & -1 \end{vmatrix} = 1(-1 - 0) = -1 \neq 0$$

Rank = 3, variables = 7. \therefore # of nondimensional numbers = 4.

We choose U, d and ρ as repeating variables.

Let $\Pi_1 = U^a d^b \rho^c P$

$$\therefore M^0 L^0 T^0 = (LT^{-1})^a (L)^b (ML^{-3})^c ML^2 T^{-3} = M^{1+c} L^{a+b-3c-2} T^{-a-3}$$

Solving we get $a = -3, b = -2, c = -1$

Thus $\Pi_1 = P/\rho U^3 d^2$

Let $\Pi_2 = U^a d^b \rho^c \omega$

$$\therefore M^0 L^0 T^0 = (LT^{-1})^a (L)^b (ML^{-3})^c T^{-1} = M^c L^{a+b-3c} T^{-a-1}$$

Solving we get $a = -1, b = 1, c = 0$

Thus $\Pi_2 = d\omega/U$

Let $\Pi_3 = U^a d^b \rho^c c \rightarrow \Pi_3 = U/c$

Let $\Pi_4 = U^a d^b \rho^c \mu \rightarrow \Pi_4 = \rho U d / \mu$

Therefore

$$P/\rho U^3 d^2 = f(d\omega/U, U/c, \rho U d/\mu)$$

At low speeds the power output is likely to depend most strongly on the size and angular speed on the propeller, and viscous effects; so $P/\rho U^3 d^2$ is likely to be determined mostly by $d\omega/U$ and $\rho U d/\mu$. At higher speeds compressibility effects are likely to be more important than viscous effects, so $d\omega/U$ and U/c are most important.

4

Exercise 8.2

- Given $L_m/L_p = 1/25$
 $P_m = 200 \text{ kPa}$
 $T_m = 300 \text{ K} \rightarrow \rho_m = P_m/RT_m = 2 \times 10^5 / (287)(300) = 2.32 \text{ kg/m}^3$
 $U_p = 30 \text{ km/h}$
 $U_m = ?$
 $D_m/D_p = ?$

Since the submarine would not operate near the sea surface, gravity is unimportant. Therefore C_D depends only on Re , which should be duplicated in the model test.

Equating $U_m L_m / \nu_m = U_p L_p / \nu_p$ we get
 $U_m = U_p (L_p/L_m) (\nu_m/\nu_p) = 30 (1/25) (1.5 \times 10^{-5} / 10^{-6}) = 18 \text{ km/h} = 5 \text{ m/s}$
error should be 25

Equating C_D , $D_m/\rho_m U_m^2 L_m^2 = D_p/\rho_p U_p^2 L_p^2$
 $D_m/D_p = (\rho_m/\rho_p) (U_m/U_p)^2 (L_m/L_p)^2 = (2.32/1000) (18/30)^2 (1/25)^2$
 $= 1.336 \times 10^{-6}$

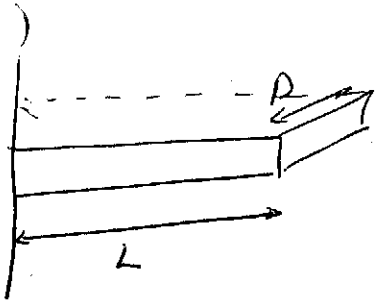
$$S_{1...} (a) \quad k D^2 \frac{\Delta T}{L^2} - h D \Delta T + q'' D^2 = 0$$

$$\text{As } L \rightarrow \infty$$

$$\frac{1}{L^2} \rightarrow 0$$

$$\therefore k D^2 \frac{\Delta T}{L^2} \rightarrow \text{negligible}$$

$$\text{as long as } \frac{D^2}{L^2} \rightarrow 0$$



$$(b) \quad h D \Delta T + q'' D^2 = 0$$

$$h D \Delta T \sim q'' D^2$$

$$\boxed{\Delta T \sim \frac{q'' D}{h}}$$

$$(c) \quad k D^2 \frac{\Delta T}{L^2} \sim q'' D^2$$

$$\boxed{L^2 \sim \frac{k \Delta T}{q''}}$$